

### Electrostatic Dust Transport On Airless Planetary Bodies

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#### Examples of Spacecraft Observations of Electrostatic Dust Transport

#### Surveyor 7: 1968-023T06:36:02



The Spokes in Saturn's B ring [Mitchell et al., Science, 2006]







Dust pond on asteroid Eros [Robinson et al., Nature, 2001]

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#### **Significance of Electrostatic Dust Transport**

Potential to explain: Surface morphology Surface porosity (thus, thermal inertia) Surface materials redistribution Space weathering

Uses in human and robotic exploration



#### **Solar Charging of Airless Bodies**



Dust particles on the regolith of airless bodies are charged and may be transported and lofted due to electrostatic forces.



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[Wang et al., GRL, 2016.]

Exposure to 120 eV electron beam

Dust particles are lofted off the surface under UV, electron beam, or plasma & electron beam conditions, in which photo- or secondary electrons are emitted.

#### **A New "Patched Charge Model"**



According to Gauss's law  $Q_b \propto (M_b - M_p) / M_{De}$  $Q_r \propto (W_r - W_b) / r$  $Q_r >> Q_b$  due to r <<  $M_{De}$ 

 $Q \boxtimes Q_r \boxtimes 0.5C(\boxtimes T_{ee}/e),$ where  $T_{ee}$  is the emitted electron temperature;  $C = 4\pi \varepsilon_0 r;$  $\mathbb{X} > 1$  (empirical constant  $4 \sim 10$ ).

[Wang et al., GRL, 2016.]

- Photo- or secondary electrons are absorbed by red surface patches in  $\bullet$ micro-cavities that are shielded from incoming photons or electrons/ions.
- These **red** patches have a very negative potential and their closeness  $\bullet$ ejects them.

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#### **Measurements of Surface Patch Potentials**



 $V_{top}$  (blue patch) is similar to  $V_{plate}$ .  $V_{bottom}$  (red patch) is negative relative to  $V_{plate}$ .

#### **Charge Measurements (Polarity)**



Negative voltage (-3 kV) grid

Positive voltage (+0.5 kV) grid

- Only negatively charged particles are accelerated under UV, electron beam, and plasma & electron beam conditions.
- This result is contrary to the generally expected positive charge due to photoemission but in agreement with our "patched charge model".

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#### **Direct Charge Measurements**



#### Procedure

- Expose dust particles to UV, electron beam, or plasma & electron beam.
- 2. Turn off the charging source.
- 3. Move the Faraday cup above the surface to accelerate charged particles to the cup where their image charges are measured.

\* The grid is used as a gate to control when charged dust will be accelerated, and also used for charge polarity measurements.

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Both irregular-shaped (Mars simulants) and microspheres (silica) are used in the measurements.

### **Charge Measurements (Magnitude)**



- Broad charge distributions due to broad size distributions of lofted dust [Wang et al., 2016].
- More irregular-shaped particles than microspheres are registered in the Faraday cup.

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#### **Charge Measurements (Magnitude)**



- Broad charge distributions due to broad size distributions of lofted dust [Wang et al., 2016].
- More irregular-shaped particles than microspheres are registered in the Faraday cup.

#### **Charge Measurements (Magnitude)**



• Charges estimated from the "patched charge model" are in a same order of magnitude with the measurements.

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#### **Lofted Particles Heights and Speeds**



- Lofted dust particles with negative charges jump higher than the predicated heights for ballistic trajectories.
- The sheath electric field changes the dust dynamics.

Size: < 44 μm in diameter <u>On Earth:</u> Initial vertical speed: 0.6 m/s, Maximum height: 1.9 cm <u>On Comet 67P:</u> Maximum height: 1,121 m



#### **Surface Mobilization of Dust Particles**

Mars simulant (38-45 µm) under plasma & electron beam (1 hour long time lapse)

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#### **Dust Spectra**



To understand the effect of surface morphology, porosity and dust size due to electrostatic dust transport on the spectra measurements

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#### **Summary and Conclusions**

- Direct charge measurements confirmed the predications of our new "patched charge model".
- Dust particles in part of a dusty surface that emits photo- or secondary electrons can attain net large negative charges, contrary to the generally expected positive charge polarity due to photoemission.
- Initial charging and launch conditions provided from our measurements are critical for dust dynamics studies and have not been well defined in the past.



### **Questions?**



#### Dust Transport in Electron Beam (120 eV)

### Fl+: +107.800 ms

Credit: Vision Research Camera: Phantom V2512

Shooting stars <u>Size</u>: < 44 μm in diameter; m/s

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Max. height: 1.5 cm; Vertical launch speed: 0.5



### Backup slides

#### Charging Mechanisms (Comparative experiments)



- Sheath electric field force is not a predominant force for dust transport. Secondary electrons (SEs) play a role in dust charging and transport.
- SE emission (SEE) from the dusty surface is smaller than from the solid surface, attributed to the absorption of emitted SEs by neighboring particles.

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#### **Examination of Current Charge Models**

• Shared charge model (uniform surface charge density)



• Charge fluctuation theory (due to discrete electron and ion fluxes to the surface)

$$\frac{\delta Q_{\rm rms}}{e} = \sqrt{\frac{CT_e}{e}},$$

<u>Case I</u>  $dQ_{rms} / Q = 807 / 1085 = 0.74$  $Q_{max} \approx 2Q$ , small enhancement.

(Sheridan and Hayes, 2011)

Charge induced by plasma is too small for dust particles to be lifted off.

#### More Plasma and Electron Beam Dust Experiments

Plasma and electron beam (120 eV)

Dust particles (Mars simulants,  $38 < d < 48 \mu$ m) in a crater 1.9 cm in diameter and 0.2 cm deep.

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#### **Surface Mobilization of Dust Particles**



#### **Charging Mechanisms (Micro-cavities)**

Potential of silica dust vs. solid surfaces



Potential on dust surface is more negative than that on solid surface due to the absorption of emitted SEs by the micro-cavities.

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#### **Trajectories of Dust Particles**



Plasma & Electron-Beam:

H<sub>meas</sub> = 2.11 – 2.73 mm (5.85 μm/pixel)

H = H<sub>meas</sub> /  $\cos\theta$  = 2.13 – 2.75 mm where,  $\theta \sim 7.24^{\circ}$  (View angle)  $v_{z,0}$  =  $(2gH)^{1/2}$  = 20.3 – 23.2 cm/s



H < 1 m

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Lunar horizon glow (Rennilson and Criswell, 1974; Colwell et al., 2007)

Our measurements on the Earth Particle diameter < 44  $\mu$ m H ≥ 0.025 m v<sub>z,0</sub> = (2gH)<sup>1/2</sup> and g<sub>moon</sub>=1/6 g<sub>earth</sub> H<sub>moon</sub> ≥ 0.15 m

#### Previous Laboratory Dust Transport Experiments (Two Examples)



Wang et al., 2010



Flanagan and Goree, 2006

What are the charging mechanisms? How big are the electrostatic forces?

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#### **Comments & Implications**

- Micron-sized insulating dust particles are recorded to jump to several centimeters high with an initial speed ~ 0.5 m/s under ultraviolet (UV) illumination or exposure to plasmas in laboratory.
- The interactions of the insulating dusty surface with UV radiation and/or plasmas are a **volume effect**, contrary to current charge models that only consider the interacting surface as a plane boundary.
- The emission and re-absorption of photo- and/or secondary electrons at the walls of micro-cavities formed between neighboring dust particles below the surface are responsible for generating unexpectedly large charges.
- Repulsive (Coulomb) force between dust particles, rather than sheath electric field force, is a dominant force to mobilize and lift dust off the surface.
- On the <u>dayside</u> surface, photoelectrons play the role. Due to much shorter UV wavelengths (i.e., higher photon energy) in space than in our laboratory, high-energy photoelectrons (> 10 eV) are expected, leading to even more negative charge on dust particles that form micro-cavities.
- On the <u>nightside</u> surface, secondary electrons play the role. Secondary electron emission from the nightside lunar surface [Halekas et al., 2009] was observed ~ 3 times smaller than that measured from a single lunar dust particle in laboratory [Horányi et al., 1998], indicating the absorption by micro-cavities.

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